NAG Fortran Library Routine Document D03PXF

Note: before using this routine, please read the Users' Note for your implementation to check the interpretation of **bold italicised** terms and other implementation-dependent details.

1 Purpose

D03PXF calculates a numerical flux function using an Exact Riemann Solver for the Euler equations in conservative form. The routine is designed primarily for use with the upwind discretisation routines D03PFF, D03PLF or D03PSF, but may also be applicable to other conservative upwind schemes requiring numerical flux functions.

2 Specification

SUBROUTINE DO3PXF(ULEFT, URIGHT, GAMMA, TOL, NITER, FLUX, IFAIL)
INTEGER

NITER, IFAIL

real

ULEFT(3), URIGHT(3), GAMMA, TOL, FLUX(3)

3 Description

D03PXF calculates a numerical flux function at a single spatial point using an Exact Riemann Solver (see Toro (1996) and Toro (1989)) for the Euler equations (for a perfect gas) in conservative form. The user must supply the *left* and *right* solution values at the point where the numerical flux is required, i.e., the initial left and right states of the Riemann problem defined below. In the routines D03PFF, D03PLF and D03PSF, the left and right solution values are derived automatically from the solution values at adjacent spatial points and supplied to the subroutine argument NUMFLX from which the user may call D03PXF. The Euler equations for a perfect gas in conservative form are:

$$\frac{\partial U}{\partial t} + \frac{\partial F}{\partial x} = 0,\tag{1}$$

with

$$U = \begin{bmatrix} \rho \\ m \\ e \end{bmatrix} \quad \text{and} \quad F = \begin{bmatrix} \frac{m}{\rho} + (\gamma - 1)\left(e - \frac{m^2}{2\rho}\right) \\ \frac{me}{\rho} + \frac{m}{\rho}(\gamma - 1)\left(e - \frac{m^2}{2\rho}\right) \end{bmatrix}, \tag{2}$$

where ρ is the density, m is the momentum, e is the specific total energy and γ is the (constant) ratio of specific heats. The pressure p is given by

$$p = (\gamma - 1)\left(e - \frac{\rho u^2}{2}\right),\tag{3}$$

where $u = m/\rho$ is the velocity.

The routine calculates the numerical flux function $F(U_L,U_R)=F(U^*(U_L,U_R))$, where $U=U_L$ and $U=U_R$ are the left and right solution values, and $U^*(U_L,U_R)$ is the intermediate state $\omega(0)$ arising from the similarity solution $U(y,t)=\omega(y/t)$ of the Riemann problem defined by

$$\frac{\partial U}{\partial t} + \frac{\partial F}{\partial y} = 0,\tag{4}$$

with U and F as in (2), and initial piecewise constant values $U = U_L$ for y < 0 and $U = U_R$ for y > 0. The spatial domain is $-\infty < y < \infty$, where y = 0 is the point at which the numerical flux is required.

The algorithm is termed an Exact Riemann Solver although it does in fact calculate an approximate solution to a true Riemann problem, as opposed to an Approximate Riemann Solver which involves some form of alternative modelling of the Riemann problem. The approximation part of the Exact Riemann

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Solver is a Newton-Raphson iterative procedure to calculate the pressure, and the user must supply a tolerance TOL and a maximum number of iterations NITER. Default values for these parameters can be chosen.

A solution can not be found by this routine if there is a vacuum state in the Riemann problem (loosely characterised by zero density), or if such a state is generated by the interaction of two non-vacuum data states. In this case a Riemann solver which can handle vacuum states has to be used (see Toro (1996)).

4 References

Toro E F (1996) Riemann Solvers and Upwind Methods for Fluid Dynamics Springer-Verlag

Toro E F (1989) A weighted average flux method for hyperbolic conservation laws *Proc. Roy. Soc. Lond.* **A423** 401–418

5 Parameters

1: ULEFT(3) – *real* array

Input

On entry: ULEFT(i) must contain the left value of the component U_i for i = 1, 2, 3. That is, ULEFT(1) must contain the left value of ρ , ULEFT(2) must contain the left value of m and ULEFT(3) must contain the left value of e.

2: URIGHT(3) - *real* array

Input

On entry: URIGHT(i) must contain the right value of the component U_i for i=1,2,3. That is, URIGHT(1) must contain the right value of ρ , URIGHT(2) must contain the right value of m and URIGHT(3) must contain the right value of e.

3: GAMMA – real Input

On entry: the ratio of specific heats γ .

Constraint: GAMMA > 0.0.

4: TOL – real Input

On entry: the tolerance to be used in the Newton-Raphson procedure to calculate the pressure. If TOL is set to zero then the default value of 1.0×10^{-6} is used.

Constraint: TOL > 0.0.

5: NITER – INTEGER

Input

On entry: the maximum number of Newton-Raphson iterations allowed. If NITER is set to zero then the default value of 20 is used.

Constraint: NITER ≥ 0 .

6: FLUX(3) - real array

Output

On exit: FLUX(i) contains the numerical flux component \hat{F}_i for i = 1, 2, 3.

7: IFAIL – INTEGER

Input/Output

On entry: IFAIL must be set to 0, -1 or 1. Users who are unfamiliar with this parameter should refer to Chapter P01 for details.

On exit: IFAIL = 0 unless the routine detects an error (see Section 6).

For environments where it might be inappropriate to halt program execution when an error is detected, the value -1 or 1 is recommended. If the output of error messages is undesirable, then the value 1 is recommended. Otherwise, for users not familiar with this parameter the recommended value is 0. When the value -1 or 1 is used it is essential to test the value of IFAIL on exit.

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Note: if the left and/or right values of ρ or p (from (3)) are found to be negative, then the routine will terminate with an error exit (IFAIL = 2). If the routine is being called from the user-supplied subroutine NUMFLX in D03PFF etc., then a **soft fail** option (IFAIL = 1 or -1) is recommended so that a recalculation of the current time step can be forced using the IRES parameter.

6 Error Indicators and Warnings

If on entry IFAIL = 0 or -1, explanatory error messages are output on the current error message unit (as defined by X04AAF).

Errors or warnings detected by the routine:

```
IFAIL = 1
```

```
On entry, GAMMA \leq 0.0, or TOL < 0.0, or NITER < 0.
```

IFAIL = 2

On entry, the left and/or right density or derived pressure value is less than 0.0.

```
IFAIL = 3
```

A vacuum condition has been detected therefore a solution can not be found using this routine. You are advised to check your problem formulation.

IFAIL = 4

The internal Newton-Raphson iterative procedure used to solve for the pressure has failed to converge. The value of TOL or NITER may be too small, but if the problem persists try an Approximate Riemann Solver (D03PUF, D03PVF or D03PWF).

7 Accuracy

The algorithm is exact apart from the calculation of the pressure which uses a Newton-Raphson iterative procedure, the accuracy of which is controlled by the parameter TOL. In some cases the initial guess for the Newton-Raphson procedure is exact and no further iterations are required.

8 Further Comments

The routine must only be used to calculate the numerical flux for the Euler equations in exactly the form given by (2), with ULEFT(i) and URIGHT(i) containing the left and right values of ρ , m and e for i = 1, 2, 3 respectively.

For some problems the routine may fail or be highly inefficient in comparison with an Approximate Riemann Solver (e.g., D03PUF, D03PVF or D03PWF). Hence it is advisable to try more than one Riemann solver and to compare the performance and the results.

The time taken by the routine is independent of all input parameters other than TOL.

9 Example

This example uses D03PLF and D03PXF to solve the Euler equations in the domain $0 \le x \le 1$ for $0 < t \le 0.035$ with initial conditions for the primitive variables $\rho(x,t)$, u(x,t) and p(x,t) given by

```
\rho(x,0) = 5.99924, \quad u(x,0) = 19.5975, \quad p(x,0) = 460.894, \quad \text{for } x < 0.5, \\
\rho(x,0) = 5.99242, \quad u(x,0) = -6.19633, \quad p(x,0) = 46.095, \quad \text{for } x > 0.5.
```

This test problem is taken from Toro (1996) and its solution represents the collision of two strong shocks travelling in opposite directions, consisting of a left facing shock (travelling slowly to the right), a right

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travelling contact discontinuity and a right travelling shock wave. There is an exact solution to this problem (see Toro (1996)) but the calculation is lengthy and has therefore been omitted.

9.1 Program Text

```
DO3PXF Example Program Text
  Mark 19 Revised. NAG Copyright 1999.
   .. Parameters ..
   INTEGER
                    NIN, NOUT
  PARAMETER
                    (NIN=5,NOUT=6)
                    NPDE, NPTS, NCODE, NXI, NEQN, NIW, NWKRES,
  INTEGER
                    LENODE, MLU, NW
                    (NPDE=3,NPTS=141,NCODE=0,NXI=0,
  PARAMETER
                    NEQN=NPDE *NPTS+NCODE, NIW=NEQN+24,
  +
                    NWKRES=NPDE * (2*NPTS+3*NPDE+32)+7*NPTS+4,
                    LENODE=9*NEQN+50, MLU=3*NPDE-1, NW=(3*MLU+1)
                    *NEQN+NWKRES+LENODE)
   .. Scalars in Common ..
                    ELO, ERO, GAMMA, RLO, RRO, ULO, URO
  real
   .. Local Scalars ..
   real
                    D, P, TOUT, TS, V
   TNTEGER
                    I, IFAIL, IND, ITASK, ITOL, ITRACE, K
   CHARACTER
                   LAOPT, NORM
   .. Local Arrays ..
                    ALGOPT(30), ATOL(1), RTOL(1), U(NPDE, NPTS),
                    UE(3,9), W(NW), X(NPTS), XI(1)
                   IW(NIW)
   .. External Subroutines .
                  BNDARY, DO3PEK, DO3PLF, DO3PLP, NUMFLX
  EXTERNAL
   .. Common blocks ..
                   /INIT/ELO, ERO, RLO, RRO, ULO, URO
   COMMON
                    /PARAMS/GAMMA
   .. Executable Statements ..
   WRITE (NOUT,*) 'DO3PXF Example Program Results'
   Skip heading in data file
   READ (NIN, *)
  Problem parameters
  GAMMA = 1.4e0
  RL0 = 5.99924e0
   RR0 = 5.99242e0
   UL0 = 5.99924e0*19.5975e0
  UR0 = -5.99242e0*6.19633e0
  ELO = 460.894e0/(GAMMA-1.0e0) + 0.5e0*RL0*19.5975e0**2
  ERO = 46.095e0/(GAMMA-1.0e0) + 0.5e0*RR0*6.19633e0**2
   Initialise mesh
   DO 20 I = 1, NPTS
      X(I) = 1.0e0*(I-1.0e0)/(NPTS-1.0e0)
20 CONTINUE
   Initial values
   DO 40 I = 1, NPTS
      IF (X(I).LT.0.5e0) THEN
         U(1,I) = RL0
         U(2,I) = UL0
         U(3,I) = EL0
      ELSE IF (X(I).EQ.0.5e0) THEN
         U(1,I) = 0.5e0*(RL0+RR0)
         U(2,I) = 0.5e0*(UL0+UR0)
         U(3,I) = 0.5e0*(ELO+ERO)
         U(1,I) = RR0
         U(2,I) = UR0
         U(3,I) = ER0
```

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```
END IF
   40 CONTINUE
      ITRACE = 0
      ITOL = 1
      NORM = '2'
      ATOL(1) = 0.5e-2
      RTOL(1) = 0.5e-3
      XI(1) = 0.0e0
      LAOPT = 'B'
      IND = 0
      ITASK = 1
      DO 60 I = 1, 30
         ALGOPT(I) = 0.0e0
   60 CONTINUE
      Theta integration
      ALGOPT(1) = 2.0e0
      ALGOPT(6) = 2.0e0
      ALGOPT(7) = 2.0e0
      Max. time step
      ALGOPT(13) = 0.5e-2
      TS = 0.0e0
      TOUT = 0.035e0
      IFAIL = 0
      CALL DO3PLF (NPDE, TS, TOUT, DO3PLP, NUMFLX, BNDARY, U, NPTS, X, NCODE,
                   DO3PEK, NXI, XI, NEQN, RTOL, ATOL, ITOL, NORM, LAOPT, ALGOPT, W,
                   NW, IW, NIW, ITASK, ITRACE, IND, IFAIL)
      WRITE (NOUT, 99998) TS
      WRITE (NOUT, 99999)
      Read exact data at output points
      DO 80 I = 1, 9
         READ (NIN,*) UE(1,I), UE(2,I), UE(3,I)
   80 CONTINUE
      Calculate density, velocity and pressure
      K = 0
      DO 100 I = 15, NPTS - 14, 14
         D = U(1,I)
          V = U(2,I)/D
          P = D*(GAMMA-1.0e0)*(U(3,I)/D-0.5e0*V**2)
          K = K + 1
          WRITE (NOUT, 99996) X(I), D, UE(1,K), V, UE(2,K), P, UE(3,K)
  100 CONTINUE
      WRITE (NOUT, 99997) IW(1), IW(2), IW(3), IW(5)
99999 FORMAT (4X,'X',6X,'APPROX D',3X,'EXACT D',4X,'APPROX V',3X,'EXAC',
+ 'T V',4X,'APPROX P',3X,'EXACT P')
99998 FORMAT (/' T = ',F6.3,/)
99997 FORMAT (/' Number of integration steps in time = ',16,/' Number ',
              'of function evaluations = ', I6,/' Number of Jacobian ', 'evaluations = ', I6,/' Number of iterations = ', I6,/)
99996 FORMAT (1X, e8.2, 6(1X, e10.4))
      SUBROUTINE BNDARY (NPDE, NPTS, T, X, U, NCODE, V, VDOT, IBND, G, IRES)
      .. Scalar Arguments ..
      real
                          Т
      INTEGER
                          IBND, IRES, NCODE, NPDE, NPTS
      .. Array Arguments ..
```

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```
real
                   G(NPDE), U(NPDE, NPTS), V(*), VDOT(*), X(NPTS)
.. Scalars in Common ..
real
                   ELO, ERO, RLO, RRO, ULO, URO
.. Common blocks ..
                    /INIT/ELO, ERO, RLO, RRO, ULO, URO
COMMON
.. Executable Statements ..
IF (IBND.EQ.O) THEN
   G(1) = \widetilde{U}(1,1) - RLO

G(2) = U(2,1) - ULO
   G(3) = U(3,1) - ELO
ELSE
   G(1) = U(1, NPTS) - RRO
   G(2) = U(2, NPTS) - URO
   G(3) = U(3, NPTS) - ERO
END IF
RETURN
END
SUBROUTINE NUMFLX(NPDE, T, X, NCODE, V, ULEFT, URIGHT, FLUX, IRES)
.. Scalar Arguments ..
real
                   T, X
INTEGER
                   IRES, NCODE, NPDE
.. Array Arguments ..
                   FLUX(NPDE), ULEFT(NPDE), URIGHT(NPDE), V(*)
.. Scalars in Common ..
real
                   GAMMA
.. Local Scalars ..
real
                   TOL
INTEGER
                   IFAIL, NITER
.. External Subroutines ..
EXTERNAL
                   D03PXF
.. Common blocks ..
COMMON
                   /PARAMS/GAMMA
.. Save statement ..
                   /PARAMS/
.. Executable Statements ..
IFAIL = 0
TOL = 0.0e0
NITER = 0
CALL DO3PXF(ULEFT, URIGHT, GAMMA, TOL, NITER, FLUX, IFAIL)
RETURN
END
```

9.2 Program Data

```
DO3PXF Example Program Data
0.5999e+01 0.1960e+02 0.4609e+03
0.5999e+01
           0.1960e+02
                       0.4609e+03
0.5999e+01 0.1960e+02 0.4609e+03
           0.1960e+02
0.5999e+01
                        0.4609e+03
0.5999e+01
            0.1960e+02
                         0.4609e+03
                       0.1692e+04
0.1428e+02
            0.8690e+01
0.1428e+02
           0.8690e+01 0.1692e+04
0.1428e+02
            0.8690e+01
                        0.1692e+04
0.3104e+02
            0.8690e+01
                         0.1692e+04
```

9.3 Program Results

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